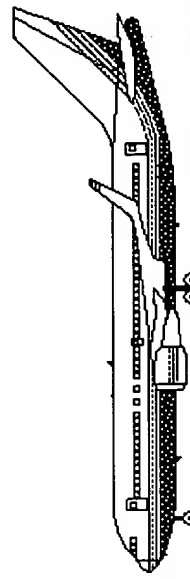
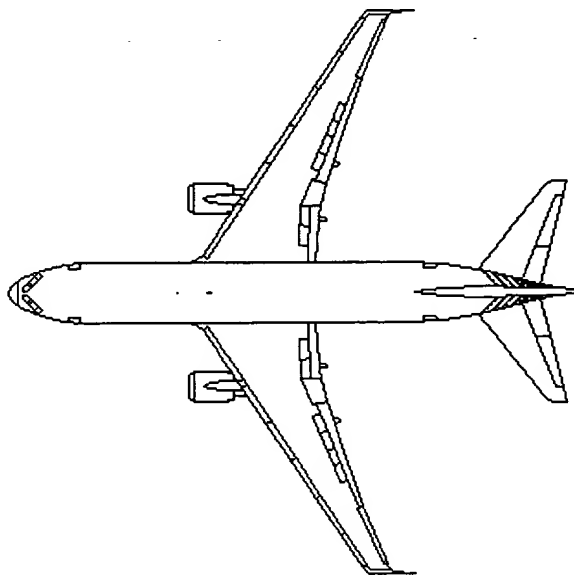
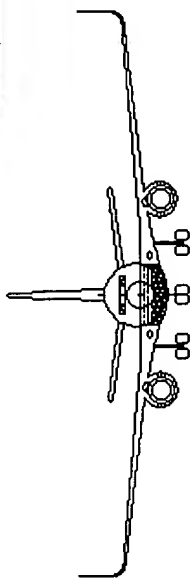


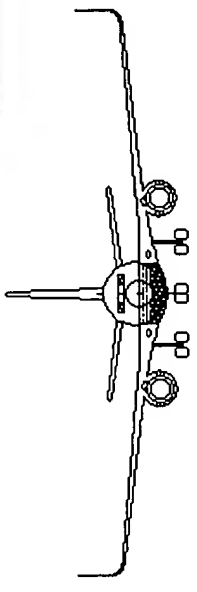
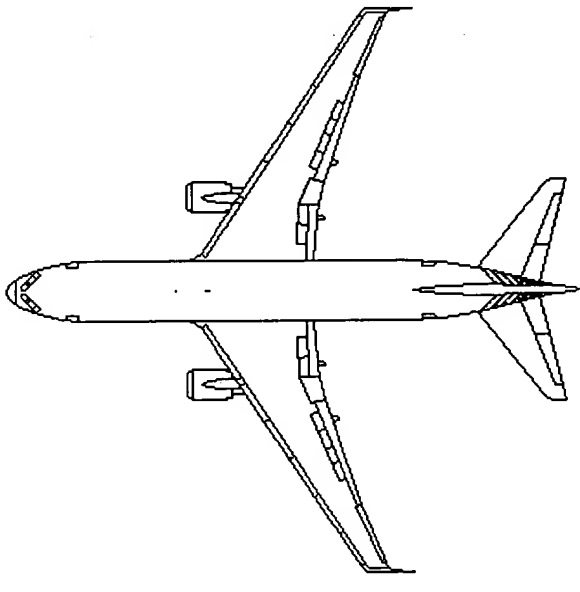
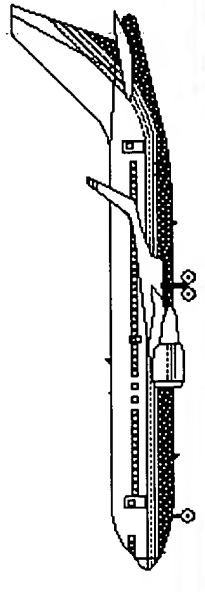
FIG. 1

Airplane Parameters

[illegible]

Airplane Parameters

Airplane application		
Maximum Takeoff Weight	300000.	LB
Number of engines	2	
Refused takeoff speed	200.	KTS
Stage Noise	3	
db delta	0.	DB
AP acoustic level	100.	DB
Max GW / Eng	150000.	LB
Max total thrust / Eng	54450.	LB
Thrust / GW ratio	0.36	
Airplane Type	Narrow Body	
Number of Passengers	221.	
Range	3370.	NM

ASSET Main Module

1

☒ TRUE

Constant

☐ FALSE

☐ FALSE

☒ TRUE ☐ FALSE

Current

154.17 ET

17.27



2

82.96

35.00 DEG

1525.58 IN

538.95 IN

ASSET EPGDS Method

FIGURE 5A









Configuration

Body CL to O/B Engine CL	0.00			IN
Side-of-Body to I/B Engine CL	122.43			IN
Side-of-Body to O/B Engine CL	0.00			IN
Dist. along LE I/B Eng. to Side-of-Body	263.32			IN
Dist along LE O/B Eng. to Side-of-Body	0.00			IN
Dist. from Fwd. E/E Bay to Front Spar BS	0.00			IN
Dist. from I/B Eng. to EE Bay	647.27			IN
Dist. from O/B Eng. to EE Bay	0.00			IN
Length of Main EE Bay	51.72			IN
H - Lower Lobe Height	56.02			IN
W1 Cabin Width	198.98			IN
W2 Cargo Floor Width	107.78			IN
Main E/E Bay Volume	257.2			FT^3
Strut location	Fan			
Accessory location	Core			

ASSET EPGDS Method

Figure 5B

AC Electrical Load Characterization

Number of Fans	6.0	
Recirculation Fans	2.0	
Number of E/E Cooling Vent Fans	2.0	
Number of E/E Cooling Supply Fans	2.0	
Number of TRUs	3.0	
Number of ACMPs	2.0	
Number of Window/Windshield Heaters	6.0	
Number of Lavatories	3.0	

Number of Wide Body Pumps	Number of Narrow Body Pumps
0.0	6.0
0.0	6.0
0.0	0.0
0.0	0.0

AC Load Summary by Flight Phase

ATA Subsystems	--- Passenger Loading ---				--- Engine Start ---				--- Taxi Out ---			
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)
21 Air Conditioning	13.72	0.82	13.72	0.82	11.32	0.82	11.32	1.00				
22 Auto Flight	0.68	0.90	0.68	0.90	0.68	0.90	0.68	0.90				
23 Communications	0.64	1.00	0.95	1.00	2.42	1.00	2.42	1.00				
24 Electrical Power	3.38	0.95	3.39	0.95	3.57	0.95	3.57	0.95				
25 Equipment/Furnishings	25.10	1.00	23.24	1.00	38.93	1.00	38.93	1.00				
26 Fire Protection	0.20	1.00	0.20	1.00	0.20	1.00	0.20	1.00				
27 Flight Control	0.07	1.00	0.07	1.00	0.07	1.00	0.07	1.00				
28 Fuel	0.00	1.00	6.08	0.83	6.08	0.83	6.08	0.83				
29 Hydraulic Power System	19.28	0.75	19.28	0.75	19.28	0.75	19.28	0.75				
30 Ice/Rain Protection	4.43	1.00	5.29	1.00	5.29	1.00	5.29	1.00				
31 Instruments	0.30	1.00	0.30	1.00	0.30	1.00	0.30	1.00				
32 Landing Gear	0.12	1.00	0.12	1.00	0.12	1.00	0.12	1.00				
Maximum Flight Phase Load	116.88	KVA<>	0.96	PF								

ASSET EPGDS Method

54

FIGURE 7A

109040" 22500660

X Exceed

File Run Goto Report

Help

AC Load Summary by Flight Phase

ATA Subsystems	---- Passenger Loading ----				---- Engine Start ----				---- Taxi Out ----			
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)
32 Landing Gear	<>	<>	0.12	1.00	<>	<>	0.12	1.00	<>	<>	0.12	1.00
33 Lights	<>	<>	10.68	1.00	<>	<>	9.71	1.00	<>	<>	9.32	1.00
34 Navigation	<>	<>	0.89	0.85	<>	<>	0.89	0.85	<>	<>	0.94	0.87
35 Oxygen	<>	<>	0.00	1.00	<>	<>	0.00	1.00	<>	<>	0.00	1.00
36 Pneumatics	<>	<>	0.00	1.00	<>	<>	0.23	1.00	<>	<>	0.00	1.00
38 Water/Waste	<>	<>	6.36	0.77	<>	<>	1.40	0.83	<>	<>	1.40	0.83
46 Electronic Library	<>	<>	0.00	1.00	<>	<>	0.00	1.00	<>	<>	0.00	1.00
49 Airplane Auxiliary Power	<>	<>	0.00	1.00	<>	<>	0.00	1.00	<>	<>	0.00	1.00
52 Doors	<>	<>	0.00	1.00	<>	<>	0.00	1.00	<>	<>	0.00	1.00
57 Folding Wing	<>	<>	0.00	1.00	<>	<>	0.00	1.00	<>	<>	0.00	1.00
73 Engine Fuel Control	<>	<>	0.00	1.00	<>	<>	0.70	0.74	<>	<>	0.70	0.74
74 Ignition	<>	<>	0.00	1.00	<>	<>	0.30	0.33	<>	<>	0.00	1.00

Maximum Flight Phase Load <> 116.88 KVA <> 0.96 PF

ASSET EPGDS Method

FIGURE 7B

X Exceed

File Run Goto Report

Help

AC Load Summary by Flight Phase

ATA Subsystems	--- Take-off & Climb ---			--- Cruise ---			--- Descent & Land ---		
	(kVA)	(PF)	(kVA)	(kVA)	(PF)	(kVA)	(PF)	(PF)	
32 Landing Gear	<> 0.12	<> 1.00	<> 0.12	<> 0.12	<> 1.00	<> 0.23	<> 1.00	<> 1.00	
33 Lights	<> 10.97	<> 1.00	<> 7.73	<> 7.73	<> 1.00	<> 11.51	<> 1.00	<> 1.00	
34 Navigation	<> 1.17	<> 0.88	<> 1.17	<> 1.17	<> 0.88	<> 1.17	<> 0.88	<> 0.88	
35 Oxygen	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
36 Pneumatics	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
38 Water/Waste	<> 0.94	<> 0.98	<> 1.14	<> 1.14	<> 0.89	<> 1.12	<> 0.94	<> 0.94	
46 Electronic Library	<> 0.00	<> 0.00	<> 0.00	<> 0.00	<> 0.00	<> 0.00	<> 0.00	<> 0.00	
49 Airplane Auxiliary Power	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
52 Doors	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
57 Folding Wing	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
73 Engine Fuel Control	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	
74 Landing Gear	<> 0.00	<> 1.00	<> 0.00	<> 0.00	<> 1.00	<> 0.00	<> 1.00	<> 1.00	

Maximum Flight Phase Load <>

116.88

KVA <>

0.96

PF

ASSET EPGDS Method

FIGURE 7C

AC Load Summary by Flight Phase

ATA Subsystems	--- Take-off & Climb ---				--- Cruise ---				--- Descent & Land ---			
	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)	(kVA)	(PF)
57 Forward wing	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
73 Engine Fuel Control	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
74 Ignition	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
75 Air	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
76 Engine Controls	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
77 Engine Indicating	0.02	1.00	0.02	1.00	0.02	1.00	0.02	1.00	0.02	1.00	0.02	1.00
78 Exhaust	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
79 Oil	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
80 Starting	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00	0.00	1.00
Flight Phase Subtotals	101.63	0.96	100.16	0.96	65.71	0.93	65.71	0.93	65.71	0.93	65.71	0.93
Error/Growth Factor (15%)	15.24	0.96	15.02	0.96	9.86	0.93	9.86	0.93	9.86	0.93	9.86	0.93
Flight Phase Totals	116.88	0.96	115.19	0.96	75.57	0.93	75.57	0.93	75.57	0.93	75.57	0.93
Maximum Flight Phase Load	116.88	KVA <>	0.96	PF								

ASSET EPGDS Method

FIGURE 7D

Essential AC Loads

	Quantity	Load per Unit	Totals
Number of Upper Recirculating Fans	2.0	1.28 KVA	15.38 KVA
Number of Lower Recirculating Fans	0.0	1.98 KVA	
Number of E/E Cooling Supply Fans	2.0	3.20 KVA	
Number of E/E Cooling Vent Fans	2.0	3.20 KVA	
Number of Hydraulic ACMP Pumps	2.0	6.41 KVA	23.30 KVA
Number of Fuel Boost Pumps	6.0	1.75 KVA	
Number of Fuel Override Pumps	0.0	4.66 KVA	
Baseline Flight & Electronics, Ice & Rain		6.75 KVA	7.08 KVA
Baseline Flight & Electronics, Electronics		6.35 KVA	13.10 KVA
Subtotal of Essential Loads			58.86 KVA
General Feeder Loss			4.12 KVA
Total of Essential Loads			62.98 KVA

ASSET EPGDS Method

File		Run		Goto		Report		Help	
<div style="display: flex; justify-content: space-between;"> Next Previous </div>									
<div style="display: flex; justify-content: space-between;"> Back Airplane Parameters </div>									
<div style="display: flex; justify-content: space-between;"> NACELLE Weight </div>									
<div style="display: flex; justify-content: space-between;"> EPGDS Configuration </div>									
<div style="display: flex; justify-content: space-between;"> Leading Edge Loads </div>									
<div style="display: flex; justify-content: space-between;"> Floor Beam Architecture </div>									
<div style="display: flex; justify-content: space-between;"> Ref sed takeoff sp Generation </div>									
<div style="display: flex; justify-content: space-between;"> Stage Noise Distribution </div>									
<div style="display: flex; justify-content: space-between;"> System Attributes Standby DC Loads </div>									
<div style="display: flex; justify-content: space-between;"> Weight Summaries IFE </div>									
<div style="display: flex; justify-content: space-between;"> AP acoustic level 0. </div>									
<div style="display: flex; justify-content: space-between;"> Max GW / Eng 100. </div>									
<div style="display: flex; justify-content: space-between;"> Max total thrust / Eng 150000. </div>									
<div style="display: flex; justify-content: space-between;"> Thrust / GW ratio 54450. </div>									
<div style="display: flex; justify-content: space-between;"> Airplane Type 0.36 </div>									
<div style="display: flex; justify-content: space-between;"> Number of Passengers Narrow Body </div>									
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<div style="display: flex; justify-content: space-between;"> 3370. </div>									

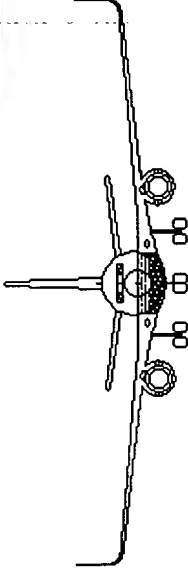
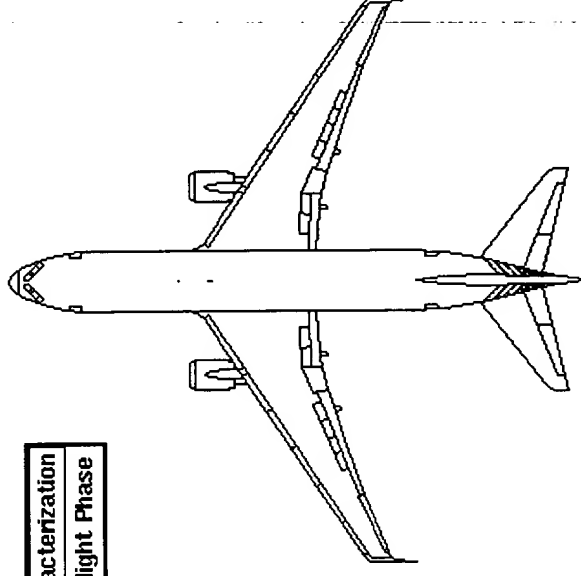
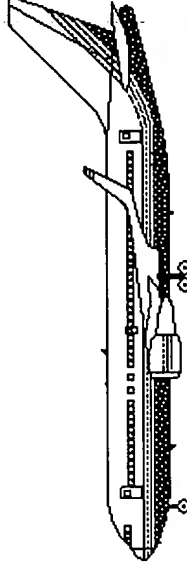




Figure 9

DC Electrical Load Characterization

Number of Main Landing Gear Wheels	4.0	
Number of APU Generators	1.0	
Number of Doors	6.0	
Number of Tanks	3.0	

ASSET EPGDS Method

Figure 10

ATA Subsystems

ATA Subsystems	Pass		Engine		Take-Off		Descent	
	Loading	(Amps)	Start	(Amps)	Taxi-Out	(Amps)	Cruise	(Amps)
21 Air Conditioning	<<	13.40	<<	14.60	<<	15.01	<<	15.14
22 Auto Flight	<<	9.83	<<	9.83	<<	9.83	<<	10.75
23 Communications (IFE, AVOD)	<<	6.74	<<	6.56	<<	7.69	<<	7.13
24 Electrical Power	<<	9.36	<<	2.27	<<	2.27	<<	2.27
25 Equipment/Furnishings	<<	18.22	<<	18.04	<<	18.04	<<	18.04
26 Fire Protection	<<	0.54	<<	0.54	<<	0.54	<<	0.54
27 Flight Control	<<	0.07	<<	0.07	<<	0.07	<<	0.07
28 Fuel	<<	6.51	<<	1.21	<<	1.21	<<	1.21
29 Hydraulic Power System	<<	0.80	<<	0.80	<<	0.80	<<	0.80
30 Ice/Rain Protection	<<	3.20	<<	5.30	<<	12.42	<<	5.40
31 Instruments	<<	36.81	<<	36.70	<<	36.70	<<	36.70
Maximum Flight Phase Direct Current Load		139.90	AMPS					

ASSET EPGDS Method

DC Load Summary by Flight Phase

ATA Subsystems	Pass		Engine		Taxi-Out		Take-Off & Climb		Cruise		Descent & Land	
	(Amps)		(Amps)		(Amps)		(Amps)		(Amps)		(Amps)	
31 Instruments	<>	36.81	<>	36.70	<>	36.70	<>	36.70	<>	36.70	<>	36.70
32 Landing Gear	<>	3.69	<>	3.69	<>	3.67	<>	3.59	<>	3.59	<>	4.07
33 Lights	<>	15.84	<>	15.77	<>	16.38	<>	19.47	<>	17.07	<>	16.05
34 Navigation	<>	1.99	<>	1.95	<>	2.45	<>	2.45	<>	2.45	<>	2.45
35 Oxygen	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
36 Pneumatics	<>	4.07	<>	4.07	<>	4.07	<>	4.07	<>	4.07	<>	4.07
38 Water/Waste	<>	2.07	<>	1.53	<>	1.53	<>	2.07	<>	1.65	<>	2.07
46 Electronic Library	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
49 Airplane Auxiliary Power	<>	1.20	<>	1.20	<>	1.20	<>	1.20	<>	0.00	<>	0.00
52 Doors	<>	1.00	<>	1.50	<>	1.50	<>	1.50	<>	1.50	<>	1.50
57 Folding Wing	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00	<>	0.00
Maximum Flight Phase Direct Current Load		139.90										

ASSET EPGDS Method

Figure 11B

ATA Subsystems

ATA Subsystems	Pass		Engine		Take-Off		Descent	
	Loading	Start	Taxi-Out	Cruise	& Land			
	(Amps)	(Amps)	(Amps)	(Amps)	(Amps)			
52 Doors	1.00	1.50	1.50	1.50	1.50	1.50	1.50	
57 Folding Wing	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
73 Engine Fuel Control	0.00	0.07	0.37	0.37	0.37	0.37	0.37	
74 Ignition	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
75 Air	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
76 Engine Controls	1.12	1.12	0.65	0.65	0.65	0.65	0.65	
77 Engine Indicating	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
78 Exhaust	0.00	0.00	0.00	0.00	0.00	0.00	0.60	
79 Oil	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
80 Starting	0.00	3.20	0.00	0.00	0.00	0.00	0.00	
Flight Phase Totals	130.46	130.02	136.40	129.41	139.90	129.41	139.53	

Maximum Flight Phase Direct Current Load

AMPS

ASSET EPGDS Method

FIGURE 11C

FileRunGotoReport

Exceed

Help

Standby DC Loads

Emergency/Standby Load

81.59

AMPS

60

ASSET EPGDS Method

FIGURE 12

IFE

Technology Era Constant	1.000	
System Factor	0.700	
Airflow Constant	156.0	CFM/KVA
Fan performance coefficient	0.00196522	KVA/CFM
IFE Power Factor	0.98	
IFE Utilization Factor	100.0	
IFE Load	0.0	KVA

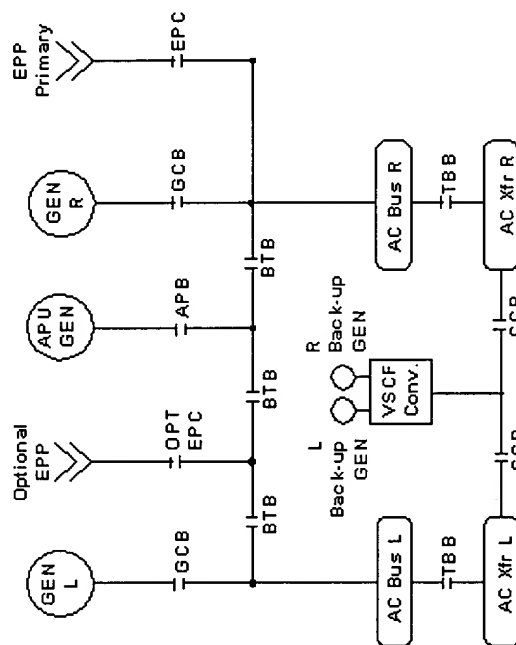
ASSET EPGDS Method

62

FIGURE 13

Main/Backup AC System

Twin, Fly-by-Wire, Isolated Architecture



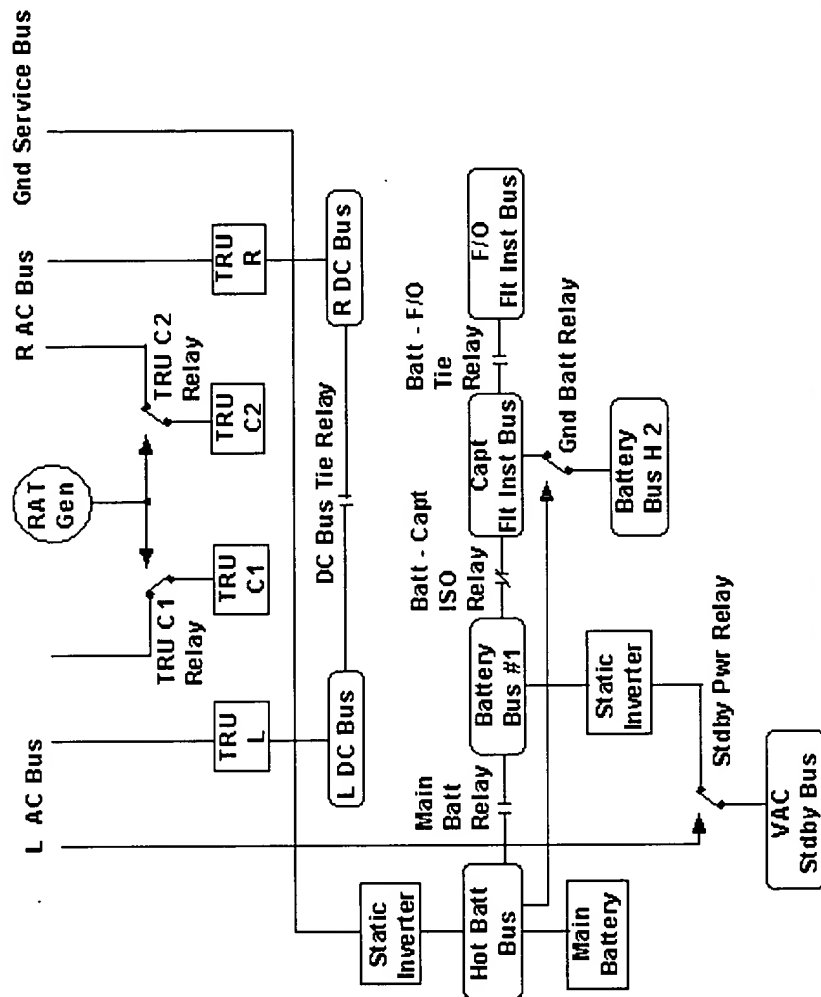
If selected, HMG Backup System not reflected in graphics.

Main Generator Type	IDG		In-flight operable	
Optional External Power Panel	<input checked="" type="checkbox"/>	TRUE	VSCF : Stand-Alone Converter	
Number of Generators per Engine	1		<input type="checkbox"/> FALSE	

ASSET EPGDS Method

Figure 15

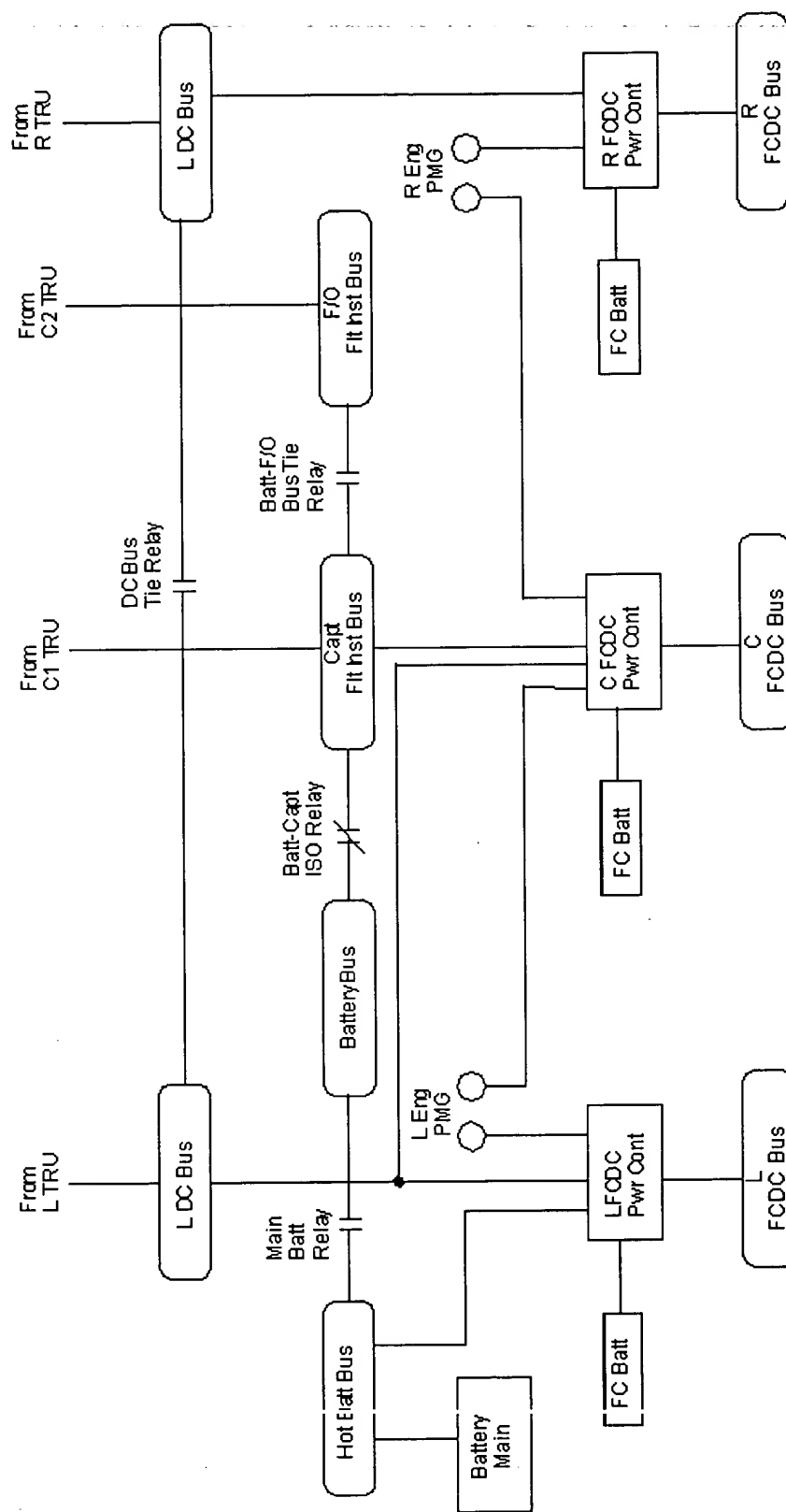
Twin, Fly-by-Wire



ASSET EPGDS Method

Figure 16

Twin, Fly-by-Wire



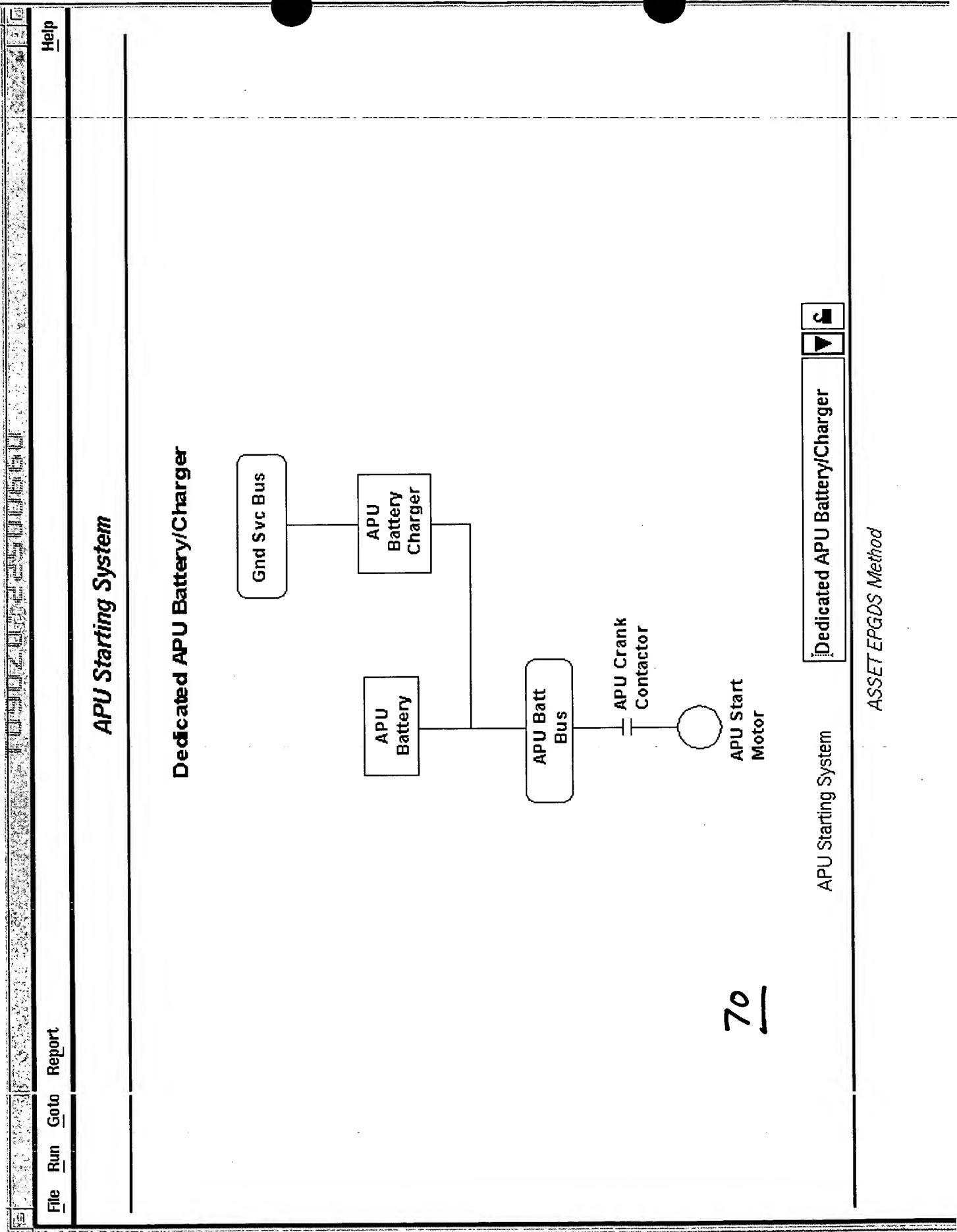
PMGs for FCDC?

TRUE

ASSET EPGDS Method

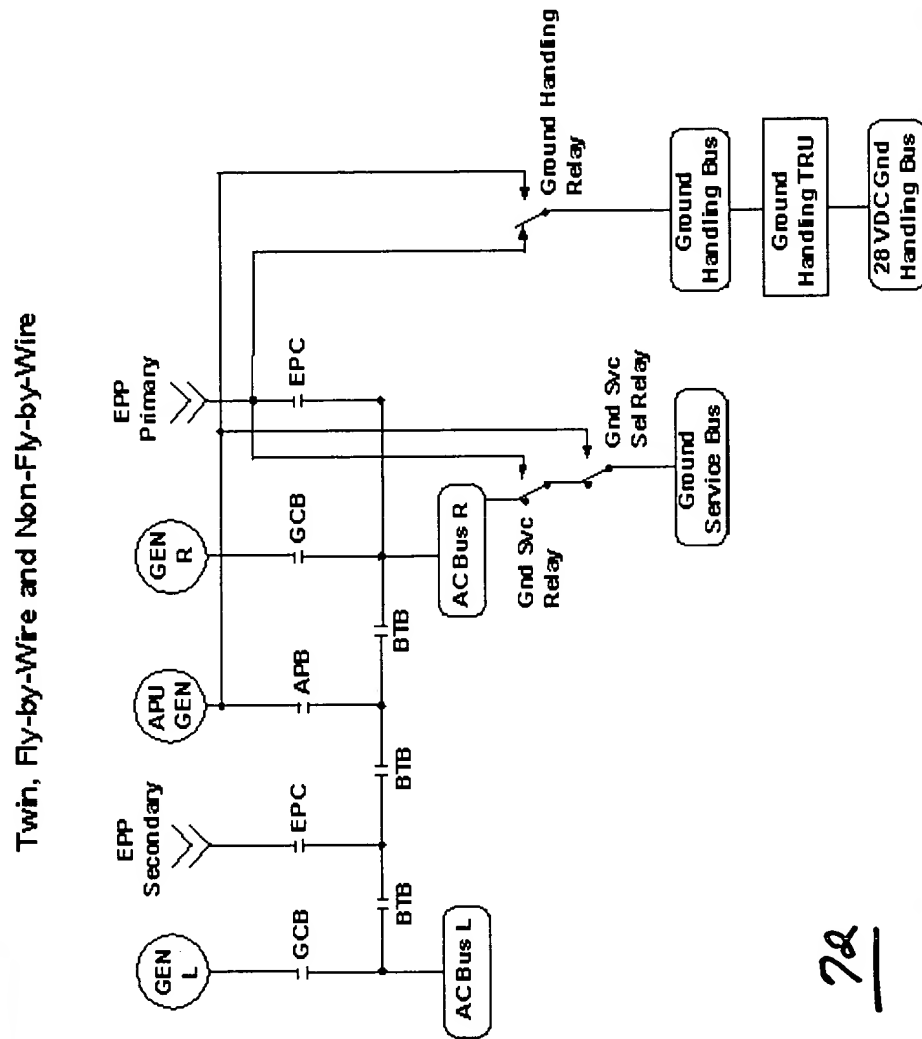
89

FIGURE 17



70

Figure 18

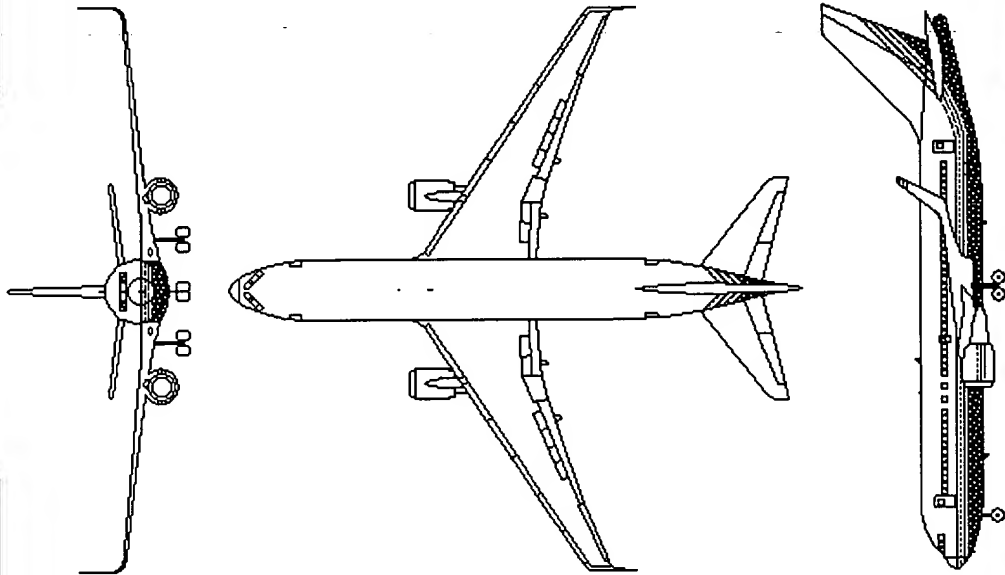


72

ASSET EPGDS Method

FIGURE 19

File		Run		Goto		Report		Help	
		Next		Previous:					
		Back							
		Airplane Parameters							
		NACELLE		On					
		EPGDS		Weight		300000.		LB	
		Leading Edge		Configuration					
		Floor Beam		Loads		2			
				Architecture					
		Refused takeoff speed		Generation		500		CLMTC	
		Stage Noise		Distribution					
		db delta		System Attributes					
				Weight Summaries					
		AP acoustic level							
		Max GW / Eng							
		Max total thrust / Eng							
		Thrust / GW ratio				0.36			
		Airplane Type				Narrow Body			
		Number of Passengers				221.			
		Range				3370.		NM	

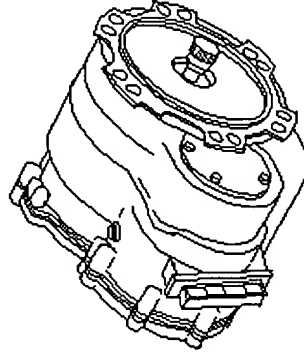


The diagram shows three views of an airplane: a top-down view, a side profile view, and a front view. The top-down view shows the wings, engines, and tail. The side profile view shows the fuselage, wings, and tail. The front view shows the nose, cockpit, and engines.

FIGURE 20

AC Power Generation

Generator Input Speed	24000.	RPM
Method of Cooling	Oil	
Generator Capacity	90.0	KVA
Main AC Power Generator Weight	110.5	LB
VSCF Converter Config.	None	
Maximum Converter Load	0.0	KVA
Main Converter Unit Weight	0.0	LB



60

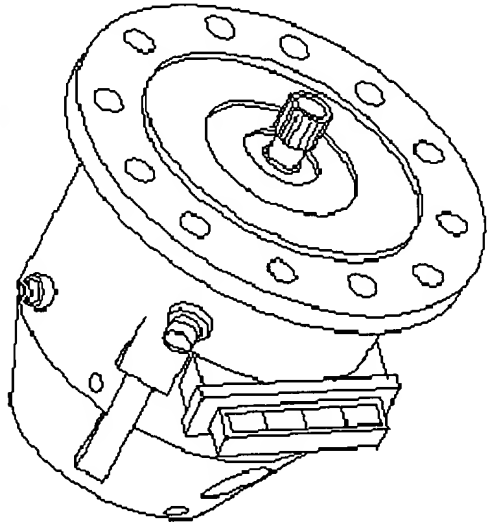
[illegible]

ASSET EPGDS Method

17

Figure 21

APU Generator



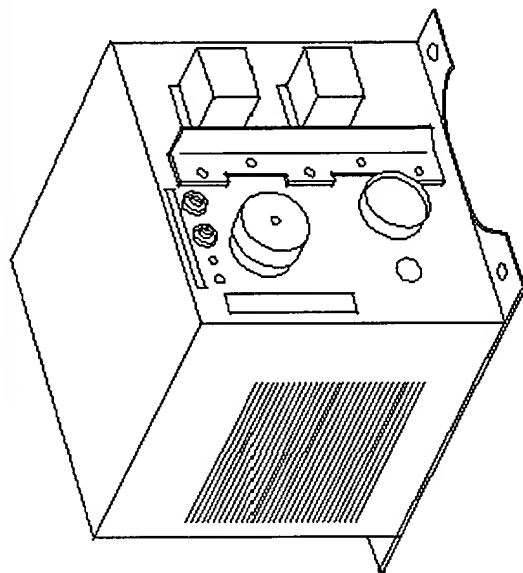
In-Flight Operable APU	<input checked="" type="checkbox"/>	TRUE	<input type="checkbox"/>
APU Generator Capacity	50.0		KVA
APU Generator Weight	64.2		LB
Number of APU Generators	1.0		







76

ASSET EPGDS Method

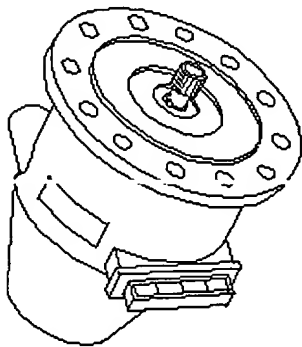
FIGURE 22

Generator Control Units

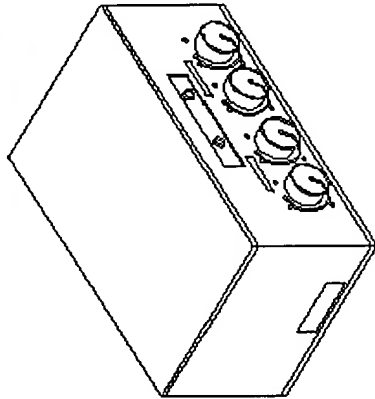


	Main AC		APU		RAT	
Unit Size	3.0		3.0		3.0	
Unit Weight	5.0		5.0		5.0	

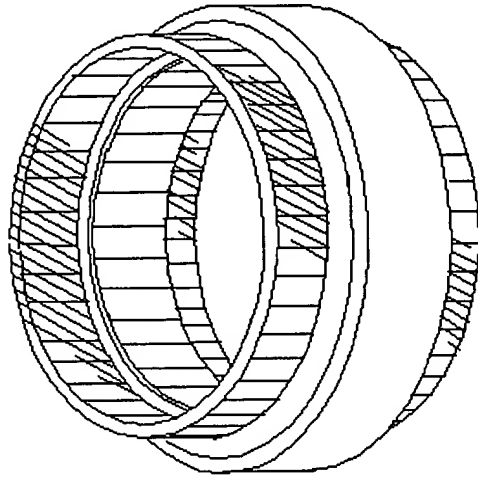
Back Up AC Power



VSCF



PMGS



Generator Type

vSCF System

Capacity

18.8

KVA

Cooling Method

Air

Input speed

12000.0

REP

Generator Weight

38.1

Number/Engine

...

PMG Configuration

Integrated with Back Up Generator

▶

Converter Configuration

Stand Alone Converter



Converter Weight

43.1



PMG Unit Weight

5.

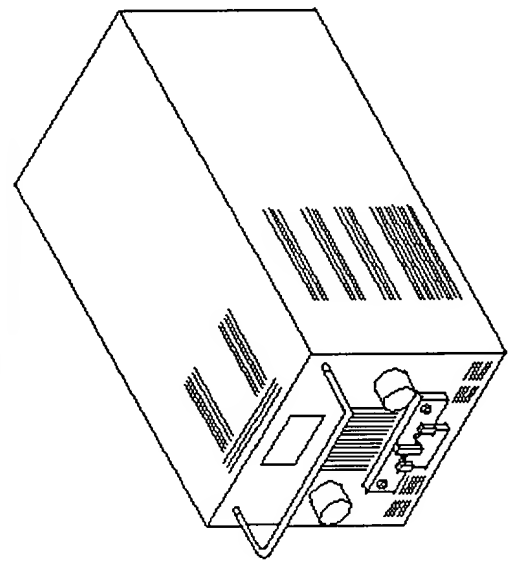
2

ASSET EPGDS Method

88

FIGURE 24

Transformer Rectifier Unit (TRU)

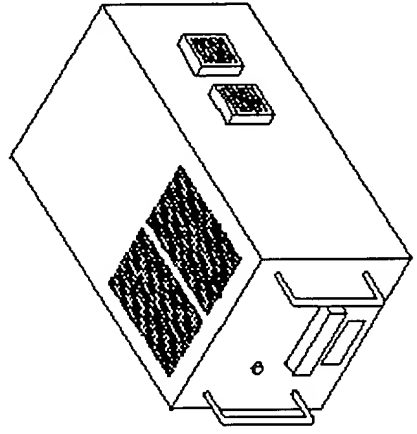
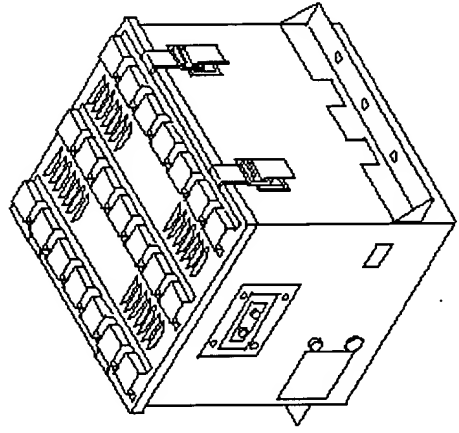


Number of TRUs	3.0	
DC Output	150.0	AMPS
Efficiency	82.0	%
TRU Weight	10.7	LB

ASSET EPGDS Method

Figure 25

Batteries and Battery Chargers



Batteries

Battery Chargers

MAIN Battery

Nominal Capacity	47.0	AMP-HRS	Output Capacity	60.0	AMPS
Battery Weight	106.0	LB	Battery Charger Weight	13.0	LB

APU Battery

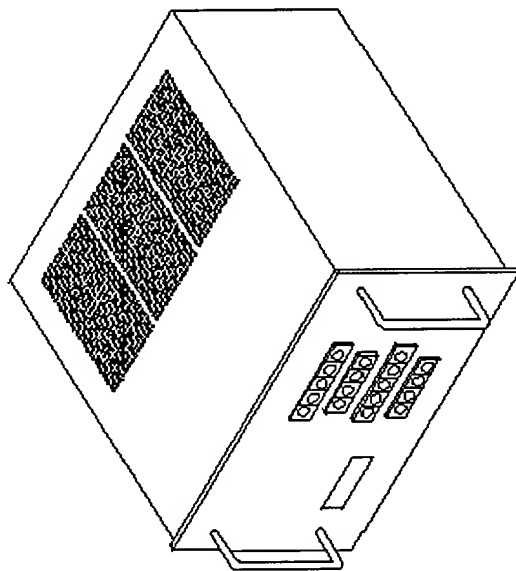
Nominal Capacity	47.0	AMP-HRS	Output Capacity	60.0	AMPS
Battery Weight	106.0	LB	Battery Charger Weight	13.0	LB

ASSET EPGDS Method

FIGURE 26

Flight Control DC Power

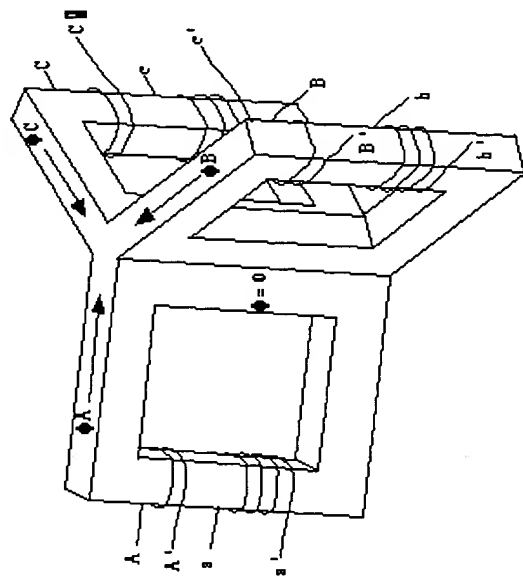
Power Supply Assemblies (PSAs)



Output Power	900.0	WATTS	Number of Dedicated Batteries	3	LB
Converter Architecture	Dual Converter		PSA Battery Unit Weight	14.3	LB
PSA Cabinet Weight	35.0	LB			

ASSET EPGDS Method

FIGURE 27



Step-Up Transformer Capacity

KVA

Step-Up Transformer

87

Step-Down Transformer Capacity

2 KVA

Step Down Transformer Weight

87

ASSET EPGDS Method

FIGURE 28

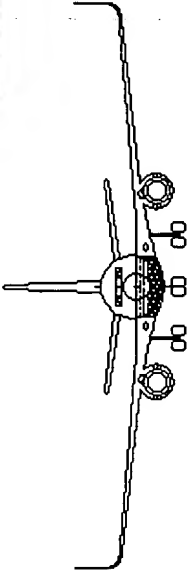
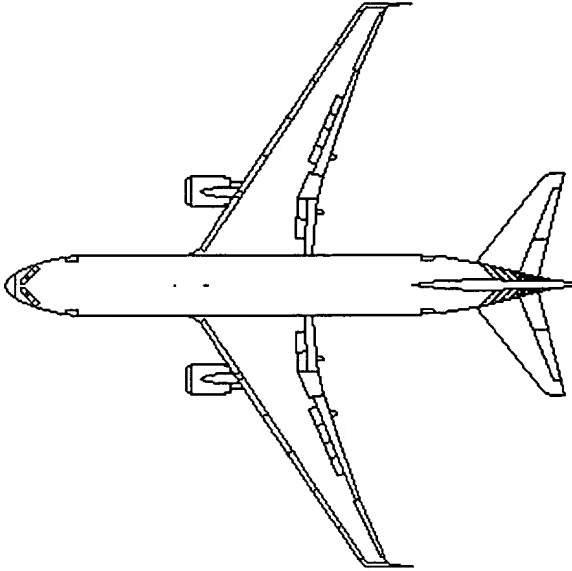
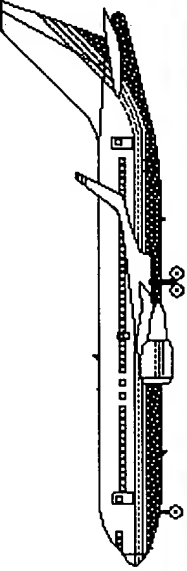
File		Run		Goto		Report		Help	
				Next		Previous			
				Back					
		Airplane Parameters							
		NACELLE							
		EPGDS							
		Leading Edge							
		Floor Beam							
		Refused takeoff speed							
		Stage Noise							
		db delta							
		AP acoustic level							
		Max GW / Eng							
		Max: total thrust / Eng							
		Thrust / GW ratio							
		Airplane Type							
		Number of Passengers							
		Range							

Airplane Parameters

Weight	300000.	LB
Configuration	2	
Loads	200.	KTS

Feeder Configuration	
Feeder Diagram	
Feeder Analysis	
Wire Type & Weight	
Panel Technology Selection	

Max GW / Eng	150000.	LB
Max: total thrust / Eng	54450.	LB
Thrust / GW ratio	0.36	
Airplane Type	Narrow Body	
Number of Passengers	221.	
Range	3370.	NM










Feeder Configuration

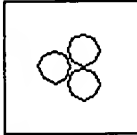
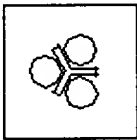
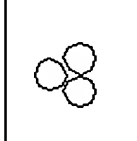
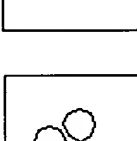
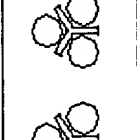
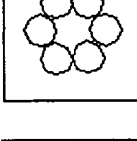
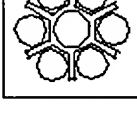
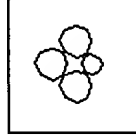
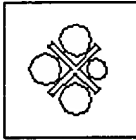
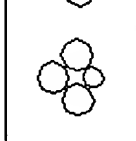
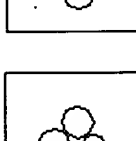
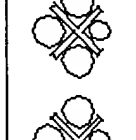
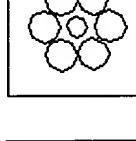
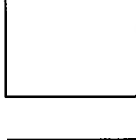
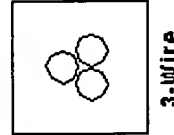
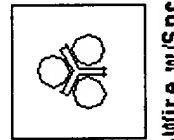
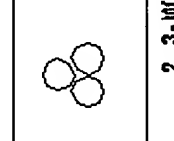
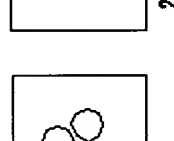
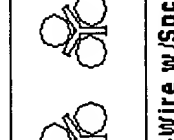
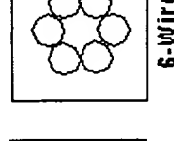
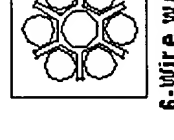
Show Data for:

MAIN



<i>Feeder 1:</i>	<>	3-Wire w/Ntrl	▼	
<i>Feeder 2:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 3:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 4:</i>	<>	2 3-Wire w/Ntrl	▼	
<i>Feeder 5:</i>	<>	None	▼	

Bundle Cross-Sections

	3-Wire		3-Wire w/Spcl		2 3-Wire		2 3-Wire w/Ntrl		2 3-Wire w/Ntrl w/Spcl		6-Wire w/Ntrl		Blank
	3-Wire		3-Wire w/Spcl		2 3-Wire		2 3-Wire w/Ntrl		2 3-Wire w/Ntrl w/Spcl		6-Wire w/Ntrl		Blank
	6-Wire w/Spcl		6-Wire		2 3-Wire w/Spcl		2 3-Wire		2 3-Wire w/Ntrl w/Spcl		6-Wire w/Ntrl		Blank

ASSET EPGDS Method

Figure 30

Wire Type & Weight




Show Data for: MAIN

Wire Type, Feeder 1:	<>	BMS 13-60 Type 7	▼	11.2	LB
Wire Type, Neutral 1:	<>	BMS 13-60 Type 7	▼	2.4	LB
Wire Type, Feeder 2:	<>	BMS 13-60 Type 7	▼	12.5	LB
Wire Type, Neutral 2:	<>	BMS 13-60 Type 7	▼	1.7	LB
Wire Type, Feeder 3:	<>	BMS 13-35 Type 1	▼	15.2	LB
Wire Type, Neutral 3:	<>	BMS 13-35 Type 1	▼	2.3	LB
Wire Type, Feeder 4:	<>	BMS 13-35 Type 1	▼	14.7	LB
Wire Type, Neutral 4:	<>	BMS 13-35 Type 1	▼	2.3	LB
Wire Type, Feeder 5:	<>	BMS 13-60 Type 22	▼	0.0	LB
Wire Type, Neutral 5:	<>	BMS 13-60 Type 22	▼	0.0	LB
TRU Feeder Weight					6.4 LB
Total Wire Weight					68.8 LB

ASSET EPGDS Method

Panel Technology Selection

Technology Factors:

Backplane	1.00	
ELMS	1.00	
Other	1.00	

ASSET EPGDS Method

The image contains three line drawings of a commercial jet aircraft. The top drawing is a top-down view showing the wings, engines, and tail. The middle drawing is a side profile view showing the fuselage, wings, and tail. The bottom drawing is a front view showing the cockpit, engines, and landing gear.

Back	on		
Airplane Parameters	Weight	300000.	LB
NACELLE	Configuration		
EPGDS	Loads	2	
Leading Edge	Architecture		
Floor Beam	Generation	200.	KTS
Refused takeoff speed	Distribution		
Stage Noise	System Attributes		
db delta	Weight Summaries	Dependability Cost	Common
		Reliability	System
AP acoustic level		Maintainability	Fuel Cost
		100.	Spares Cost
Max GW / Eng		150000.	Line Maintenance
Max total thrust / Eng		54450.	Shop Maintenance
Thrust / GW ratio		0.36	Schedule
Airplane Type			Dependability
Number of Passengers	Narrow Body		
Range	221.		
	3370.		NM

Common Dependability Cost Inputs

Number of Main Generators per Airplane

Average Number of Flights per Year per Airplane

Average Flight Hours per Flight

Airplane Fleet Size

Length of System Life in Years (1 - 30 Yrs,)

Average Non-fuel Inflation Rate beyond Present Year

Minimum Attractive Rate of Return

2		
1100.		
3.40		HRS
30		
30		YEARS
0.035		%
0.12		%

ASSET EPGDS Method

FIGURE 36

System Acquisition Costs

System Acquisition Cost, Base Year (per fleet)	0.	DOLLARS
System Support Equipment Cost, Base Year (per fleet)	0.	DOLLARS
System Initial Training Cost, Base Year (per fleet)	0.	DOLLARS
System Acquisition Cost per Airplane per Year	4078.	DOLLARS

ASSET EPGDS Method

FIGURE 37

Fuel Costs

Fuel Cost per Gallon, Base Year	0.49	DOLLARS
Lbs Fuel Burned / Flight Hour / Lb Additional Weight	0.0310	HRS^-1
System Weight (per airplane)	221.0	LB
System Direct Horsepower Requirement (per airplane)	0.	HP
System Drag Horsepower Requirement (per airplane)	0.	HP
System Cooling Horsepower Requirement	0.	HP
System Pound of Fuel per Block Trip (per airplane)	0.	LB
Average Fuel Inflation Rate Beyond Present Year	0.035	%
Fuel Cost (NPV of Life Cycle Cost)	615800.	DOLLARS
Fuel Cost per Airplane per Year	1860.	DOLLARS

ASSET EPGDS Method

FIGURE 38

Spares Costs

Cost / Spare Unit, Base Year	270000.	DOLLARS
Spares Holding Factor	0.12	%
Shop Turnaround Time in Days	34.0	DAYS
Main Base Fill Rate (must be less than 1)	0.95	
Mean Time Between Unscheduled Removals	12000.	HRS
Mean Time Between Overhauls	0.	HRS
Number of Spares Required	5.	
Initial Spares Cost	1350000.	DOLLARS
Spares Holding Cost (NPV of Life Cycle Cost)	1787786.	DOLLARS
Spares Cost (NPV of Life Cycle Cost)	3137786.	DOLLARS
Spares Cost per Airplane per Year	9478.	DOLLARS

ASSET EPGDS Method

Figure 39

Line Maintenance Costs

Direct Labor Rate per Hour	21.00	DOLLARS/HOUR
Maintenance Labor Burden Factor	2.4	
Mean Time Between Unscheduled Removals	12000.	HRS
Line Labor Hours Required per Removal	2.0	HRS
Line Labor Hours per Maintenance Action (Non-Removal)	0.5	HRS
Maintenance Actions per 1000 Flight Hours (Non-Removal)	0.50	HRS^-1
Line Maintenance Cost (NPV of Life Cycle Cost)	73673.	DOLLARS
Line Maintenance Cost per Airplane per Year	223.	DOLLARS

ASSET EPGDS Method

Figure 40

Shop Maintenance Costs

Direct Labor Rate per Hour	21.00	DOLLARS/HOUR
Maintenance Labor Burden Factor	2.4	
Mean Time Between Unscheduled Removals	12000.	HRS
Main Generator Mean Time Between Failures	26000.	HRS
Mean Time Between Overhauls	0.	HRS
Shop Labor Man-Hours per Unconfirmed Failure (Test Time)	8.0	HRS
Shop Labor Man-Hours per Failure (Repair and Test)	48.0	HRS
Shop Labor Hours per Overhaul	0.0	HRS
Average Shop Material Cost per Failure, base year	67500.	DOLLARS
Overhaul Materials Cost per Overhaul	0.	DOLLARS
Shop Maintenance Cost (NPV of Life Cycle Cost)	6819057.	DOLLARS
Shop Maintenance Cost per Airplane per Year	20597.	DOLLARS

ASSET EPGDS Method





Scheduled Maintenance Costs






		DOLLARS/HOUR
Direct Labor Rate per Hour		\$21.00
Maintenance Labor Burden Factor		2.4
Mean Time Between Unscheduled Removals		12000. HRS
Schedule Maintenance Inspection Man Hours per 1000 Flight Hours		7.0
Rectification Man Hours per 1000 Flight Hours		0.0
Scheduled Maintenance Material Dollars per 1000 Flight Hours		0.00 HRS^-1
Scheduled Maintenance Cost (NPV of Life Cycle Cost)	\$237712.	DOLLARS
Scheduled Maintenance Cost per Airplane per Year	\$739.	DOLLARS

ASSET EPGDS Method

FIGURE 42

Schedule Interruption Costs

Average Delay Cost per Delay Hour	10300.		DOLLARS/HOUR
Average Cancellation Cost per Cancellation	51000.		
Average Air Turnback Cost per Turnback	36700.		DOLLARS
Average Diversion Cost per Diversion	43000.		DOLLARS

Number of Delays per 100 Departures	0.0030	
Average Delay Time (Hours)	1.70	HRS 
Number of Cancellations per 100 Departures	0.0001	
Number of Air Turnbacks per 100 Departures	0.0002	
Number of Diversions per 100 Departures	0.0000	

Schedule Interruptions Cost (NPV of Life Cycle Cost)	\$939999.	DOLLARS
Schedule Interruptions Cost per Airplane per Year	\$1492.	DOLLARS

ASSET EPGDS Method

FIGURE 43

Microsoft Word 2003

Dependability Cost Summary

83b

83a

	NPV of Life Cycle Cost	Per Airplane per Year	
Line Maintenance Cost	73673.	223.	DOLLARS
Shop Maintenance Cost	6819057.	20597.	DOLLARS
Scheduled Maintenance Cost	1237712.	3739.	DOLLARS
Schedule Interruptions Cost	493999.	1492.	DOLLARS
Spares Cost	3137786.	9478.	DOLLARS
Fuel Cost	615800.	1860.	DOLLARS
Dependability Cost	12378028.	37388.	DOLLARS

83

ASSET EPGDS Method

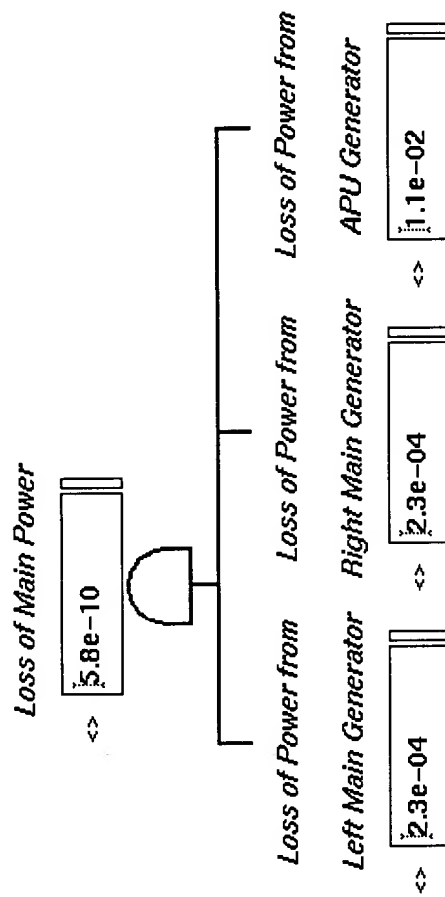
Figure 44

Reliability Inputs

Average Flight Hours per Flight		IFSD Rates (per 1000 flight hours)	
LRU MTBF's			
Main Generator MTBF	3.40	0.010	HRS^-1
APU Generator MTBF	26000.	0.200	HRS^-1
VSCF Backup Generator MTBF	20000.		
Generator Control Unit (GCU) MTBF	20000.		
Backup Converter MTBF	250000.	0.010	
Generator Control Breaker (GCB) MTBF	15000.		
	300000.	3.2e-03	
Failure to Start Probabilities			
		APU No-Start Probability	
		Probability of RAT Unavailable when Required	
Other Failure Rates (per flight hour)			
Ram Air Turbine MTBF	10000.	1.2e-05	HRS^-1
RAT Gen. Control Unit MTBF	40000.	3.0e-06	HRS^-1
Permanent Magnet Generator (PMG) MTBF	420000.	1.2e-05	HRS^-1
Main and APU Battery MTBF	25000.		
Main and APU Battery Charger MTBF	80000.		

ASSET EPGDS Method

841



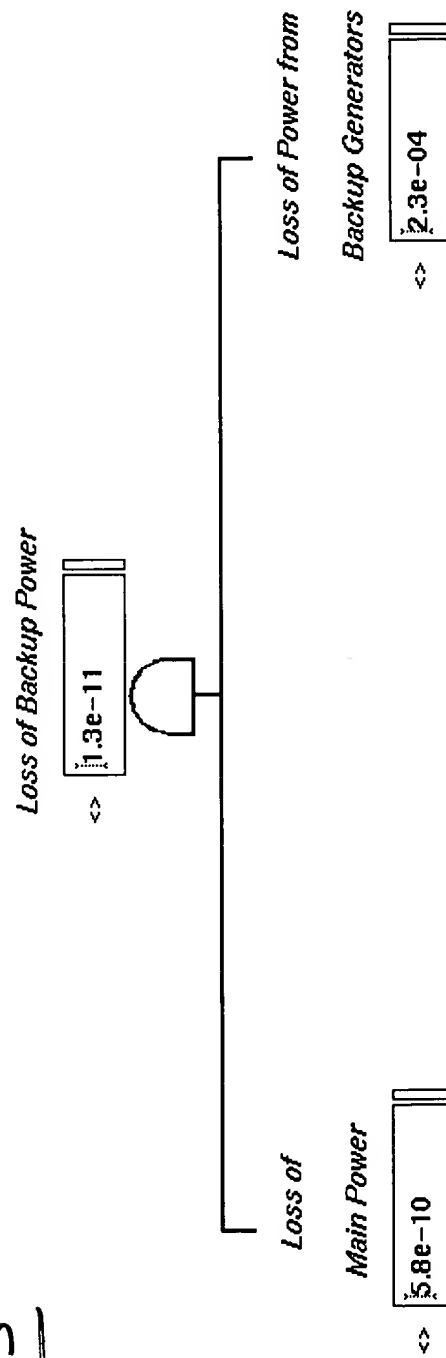
AND Gate OR Gate

***** NOTE: This is a Read-Only screen *****

FIGURE 47

Backup Power Reliability

58

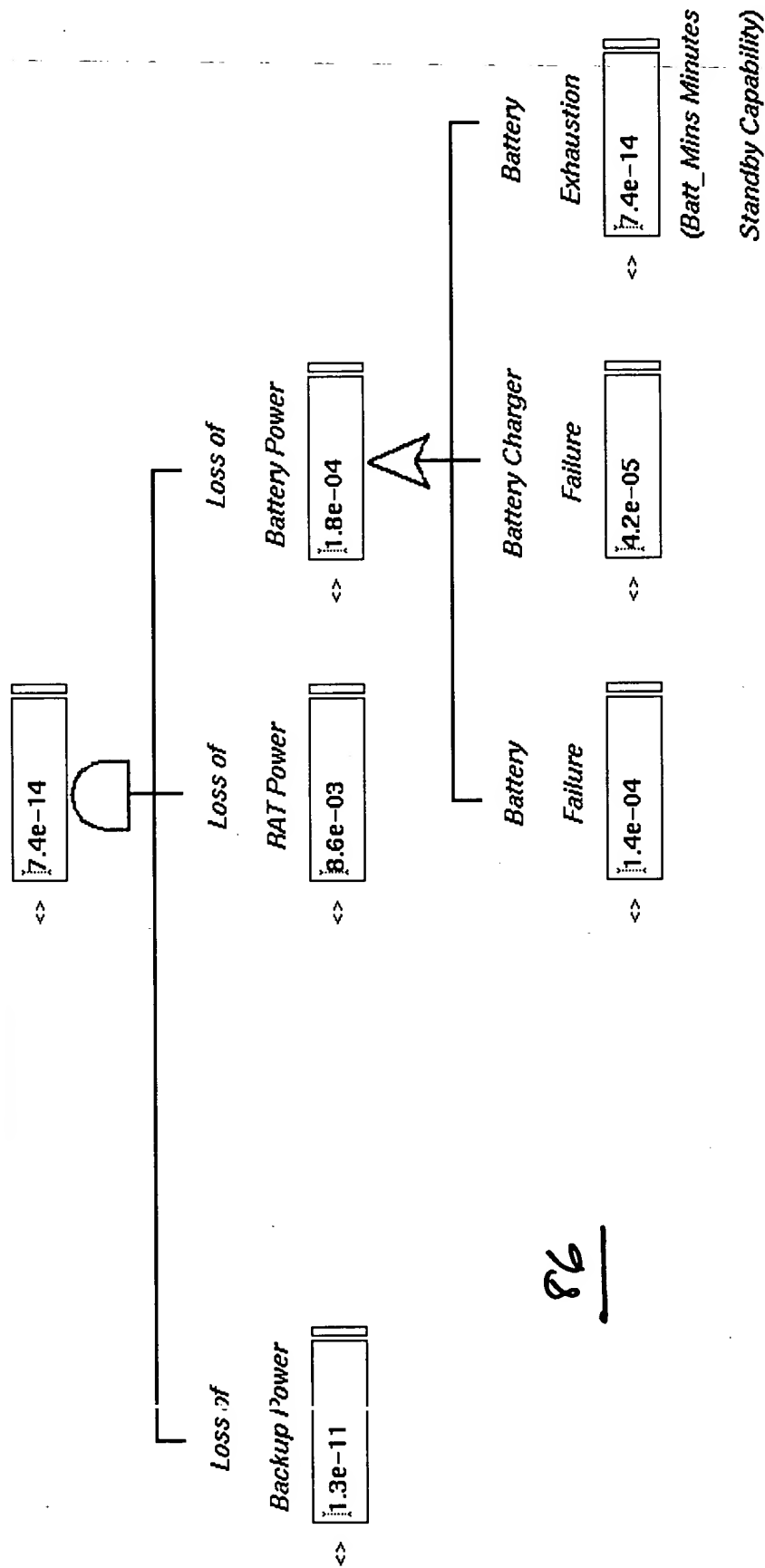


AND Gate OR Gate

***** NOTE: This is a Read-Only screen *****

FIGURE 48

Loss of Standby Power

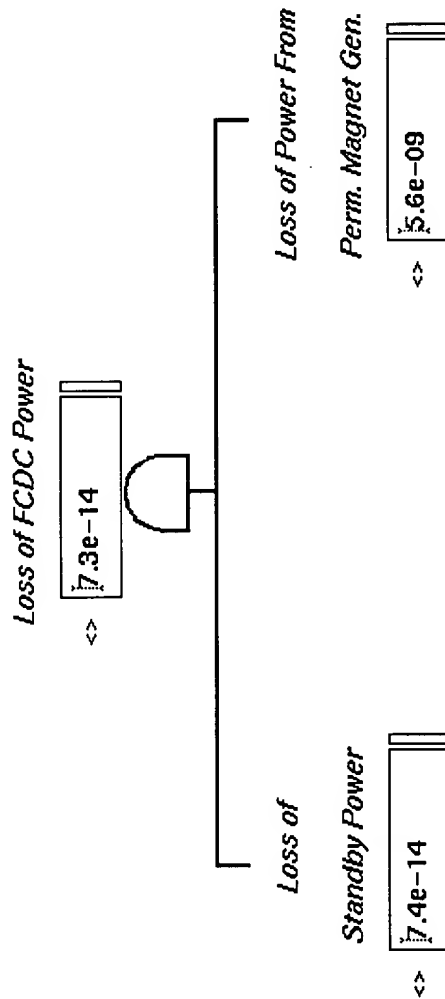


AND Gate OR Gate

***** NOTE: This is a Read-Only screen *****

FIGURE 49

FCDC Power Reliability



8.8

AND Gate OR Gate

***** NOTE: This is a Read-Only screen *****

Figure 50

Preparation Times

	Unscheduled	Servicing	Alignment & Adjustment
<i>Maintenance Preparation Times (Flight Hours)</i>			
Main Generator Unscheduled Removal Maintenance Coordination Time	0.10	0.10	0.10
Main Generator Unscheduled Removal Dispatch Delay Time	0.20		
Main Generator Unscheduled Removal Airplane Ferrying Time	1.00		
Main Generator Unscheduled Removal Supply Delay Time	0.	0.10	
Main Generator Unscheduled Removal Spares & Equipment Issuing Time	0.50		0.50
Main Generator Unscheduled Removal Transport Delay Time	0.60		
Main Generator Unscheduled Removal Maintenance Delay Time	0.70	0.10	0.10
Main Generator Unscheduled Removal Maintenance Preparation Time	4.	0.30	1.

ASSET EPGDS Method

92

Inherent Availability

Maintenance Preparation Times(Flight Hours)

Main Generator Mean Time to Repair	3.190	
Main Generator Mean Maintenance Preparation Time	0.520	
Main Generator Mean Maintenance Down Time	3.710	
Main Generator Mean Time Between Maintenance	600.0	
Main Generator Inherent Availability	0.99385	

94

Figure 54

ATA 24-09, Electrical Power Distribution

ASSET EPGDS Method

Figure 56

24-10, Generator Drive

[illegible]

ASSET EPGDS Method

Figure 57

24-21, Power and Regulation

[illegible]

ASSET EPGDS Method

ATA 24-22, Controls and Indication

ASSET EPGDS Method

FIGURE 59

ATA 24-25, Back-up Generators

ASSET EPGDS Method

FIGURE 60

24-28, Feeders

Component #	Component Designation	Quantity	Unit Wt	Subtotal
Gen_Fdrs	Main Generator Feeders, R	1	68.8	68.8
Gen_Fdrs	Main Generator Feeders, L	1	68.8	68.8
Gen_I	Main Generator Feeders Installation, R	1	12.4	12.4
Gen_I	Main Generator Feeders Installation, L	1	12.4	12.4
Gen_C	Main Generator Feeders Connectors, R	1	10.3	10.3
Gen_C	Main Generator Feeders Connectors, L	1	10.3	10.3
APU_Fdrs	APU Feeders	1	68.8	68.8
APU_C	APU Feeders Connectors	1	12.4	12.4
APU_I	APU Feeders Installation	1	10.3	10.3
		0	0.0	0.0
		0	0.0	0.0
		0	0.0	0.0
		0	0.0	0.0
ATA 24-28, Feeders				274.4

ASSET EPGDS Method

FIGURE 6)

24-31, Batteries

[illegible]

ASSET EPGDS Method

FIGURE 62

24-32, Transformer Rectifier

[illegible]

ASSET EPGDS Method

FIGURE 63

24-33, Emergency Generator

[illegible]

ASSET EPGDS Method

Figure 64

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24-40, External Power

[illegible]

ASSET EPGDS Method

Figure 6b

24-51, AC Power Distribution

[illegible]

ASSET EPGDS Method

24-60, DC Power Distribution

[illegible]

ASSET EPGDS Method

FIGURE 68

WW-01, Wiring Provision

[illegible]

ASSET EPGDS Method

FILE 69

100

100

Airplane Parameters

Airplane application

Maximum Takeoff Weight

3000000. LB

ASSET Report

FC 32 Report

Component #	Component Designation	Qty	Unit	Wt Su (LB)
32	Electrical Power Generation & Distribution System			
32-01	AC Power System	1		255.3
32-01-01	AC POWER GENERATION EQUIPMENT	1		233.9
32-01-01-01	MAIN AC POWER GENERATORS INSTLD	2		90.0
32-01-01-01-01	PRIME DRIVE GENERATOR	2		4.8
32-01-01-01-02	QUICK ATTACH DETACH (QAD)	2		13.8
32-01-01-01-03	GENERATOR FLUIDS	2		4.7
32-01-01-01-05	HARDWARE INSTALLATION	2		3.6
32-01-01-01-06	WIRING INSTALLATION	1		5.0
32-01-01-02	GENERATOR CONTROL UNITS	1		16.4
32-01-01-06	BUS POWER CONTROL UNITS	1		269.6
32-01-05	EROPS-VSCF POWER GENERATION SYSTEM	1		95.9
32-01-05-01	VSCF GENERATORS & OIL	2		38.1
32-01-05-01-01	VSCF GENERATOR	2		9.9
32-01-05-01-02	VSCF GENERATOR OIL			

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ASSET Main Module

Figure 73

Airplane Parameters

Airplane application

Maximum Takeoff Weight

300000. LB

87

CLASSIC Report

AC_stdby_Load	AC Standby Load
AGen_MTBf	APU Generator MTBF
APA	Airplane application
APUG_Cap	APU Generator Capacity
APUG_Cap_As_Built	APU Generator Capacity
APUG_Wt	APU Generator Weight
APU_Batt_Cap	Nominal Capacity
APU_Batt_Chgr_Cap	Output Capacity
APU_Batt_Chgr_Wt	Battery Charger Weight
APU_Batt_Chgr_Wt_As_Built	Built Battery Charger Weight
APU_Batt_Wt	Battery Weight
APU_Batt_Wt_As_Built	Battery Weight
APU_Ch_Prob	Probability of Loss of APU Generating Channel
APU_Feeder	APU Feeder Configuration
APU_Feeder	APU Feeder Configuration
APU_Feeder	APU Feeder Configuration
APU_Feeder	APU Feeder Configuration
APU_Feeder	APU Feeder Configuration
APU_GCU_Size	APU Generator GCU Size
APU_GCU_Wt	Unit Weight

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95

Figure 74

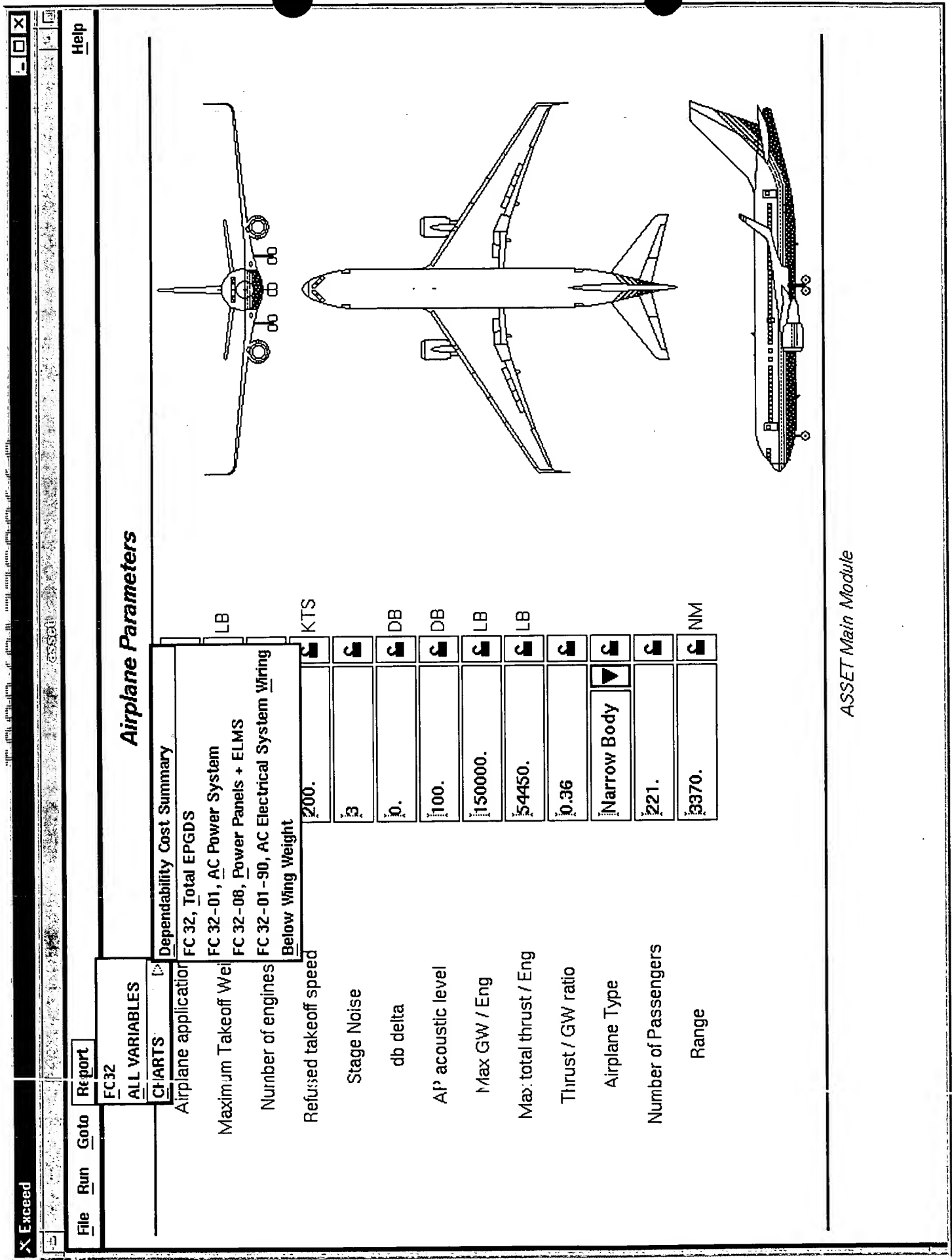
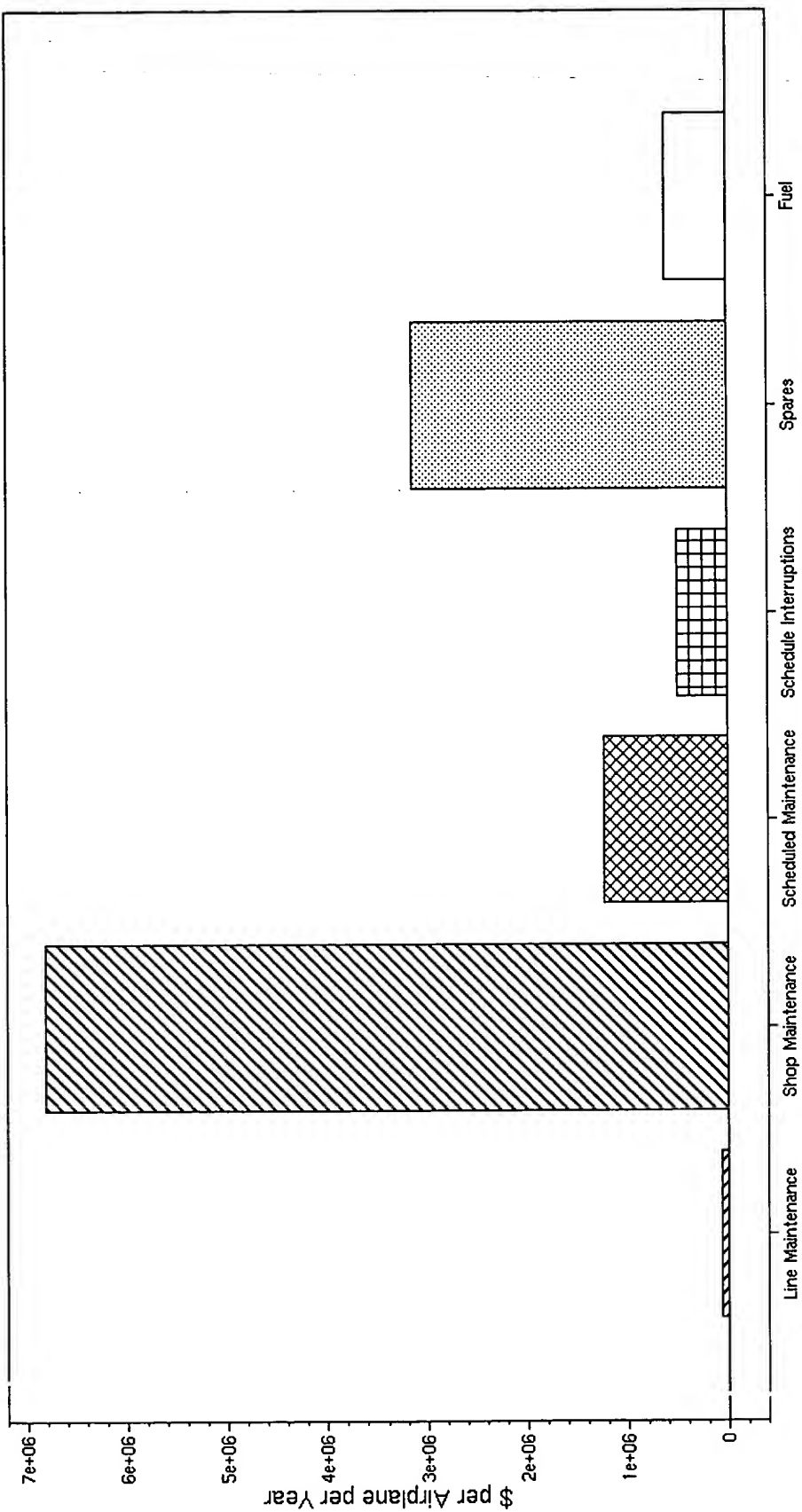


Figure 75

Dependability Cost Summary



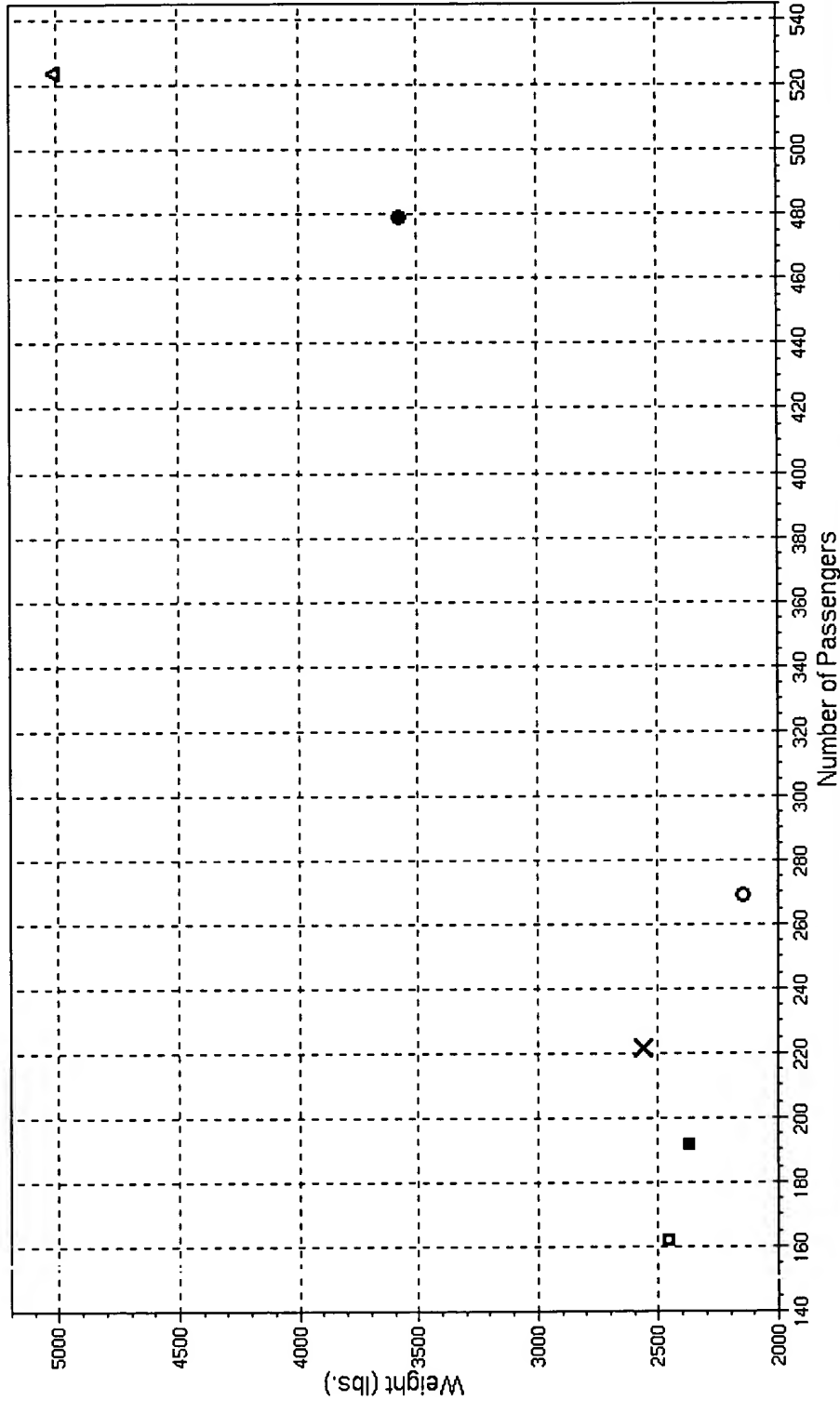
100

Return

Figure 76

asset

FC32, Total EPGDS Weight



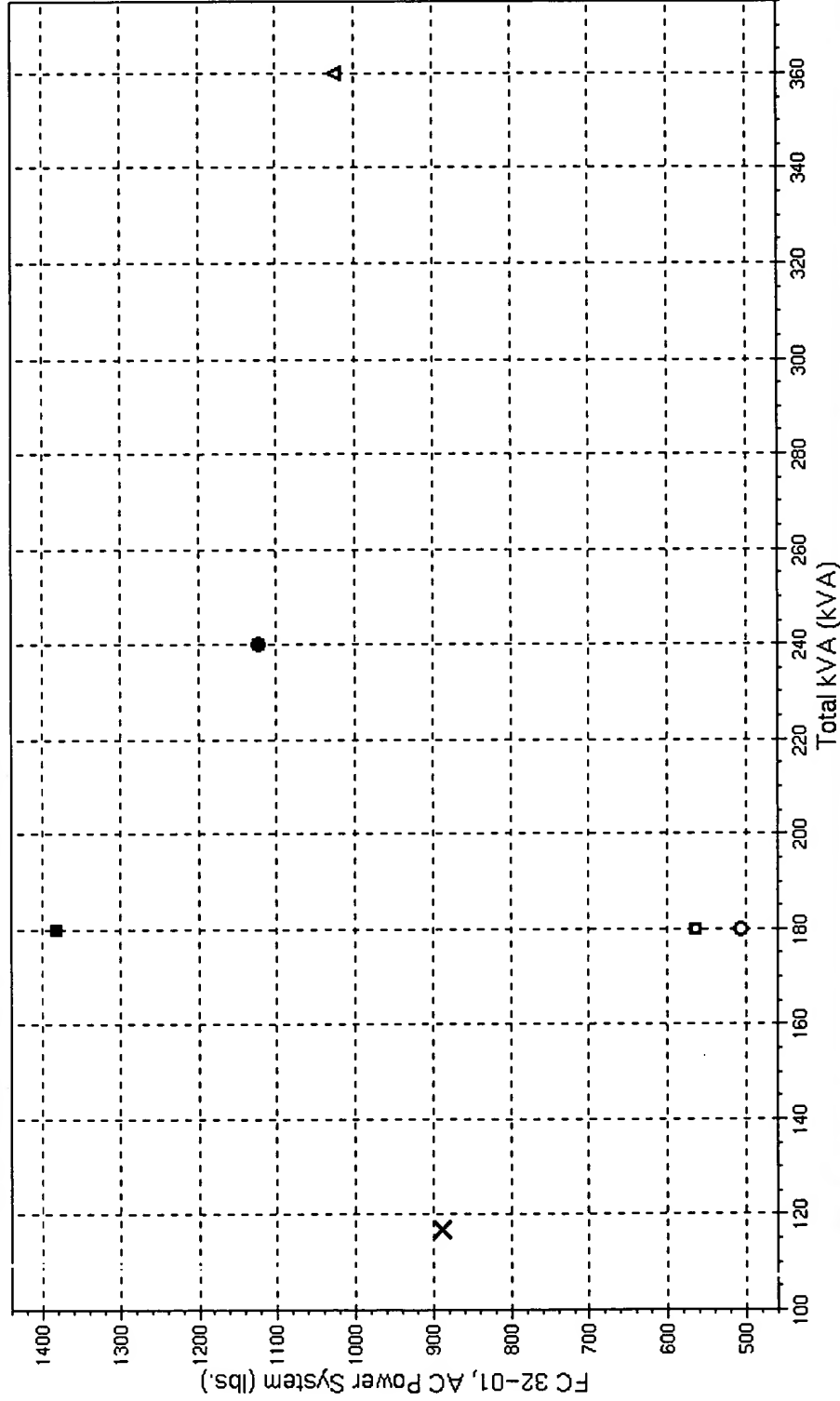
Return

102

Figure 77

ASST: Chart

FC 32-01, AC Power System Weight



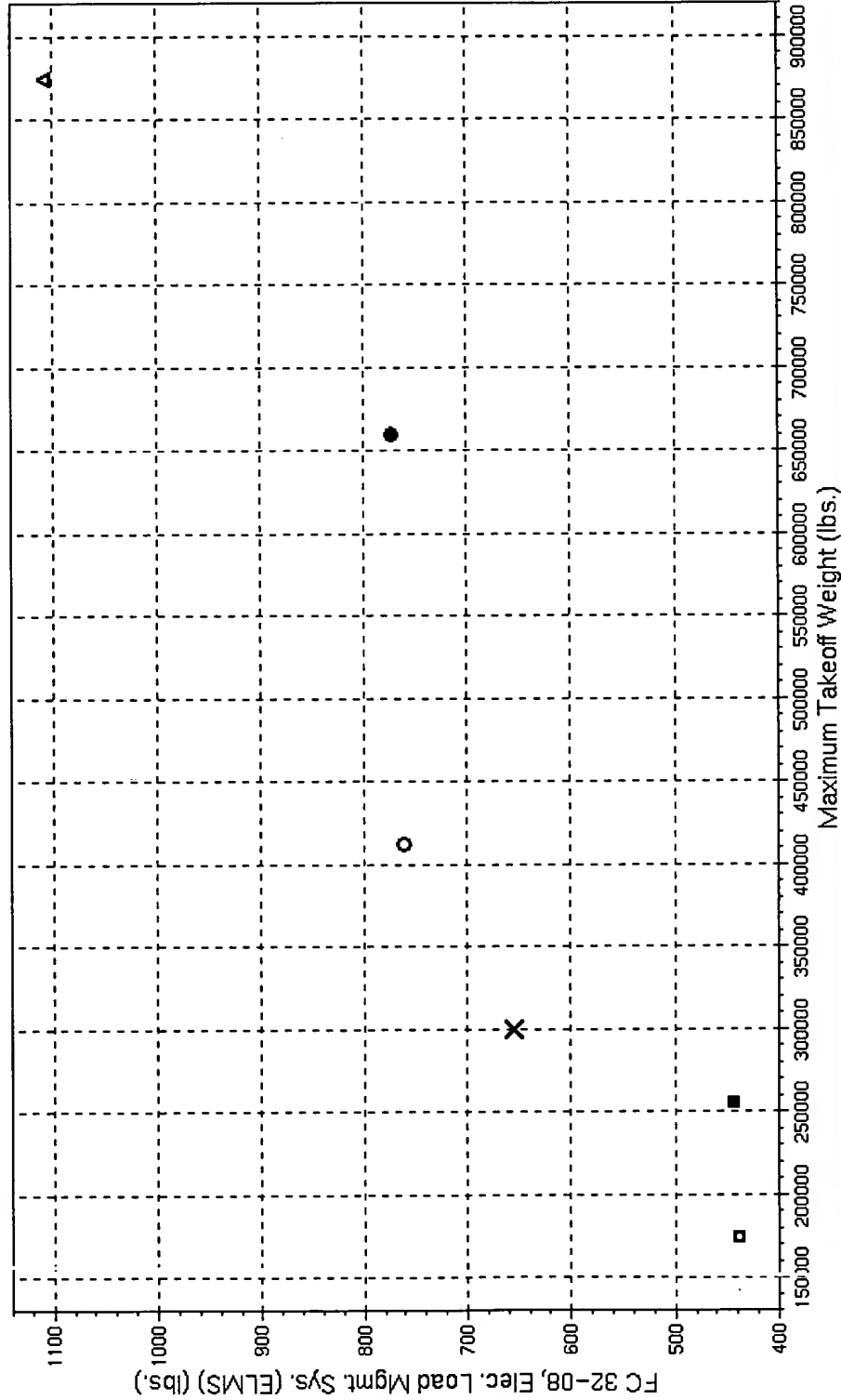
Return

104

Figure 78

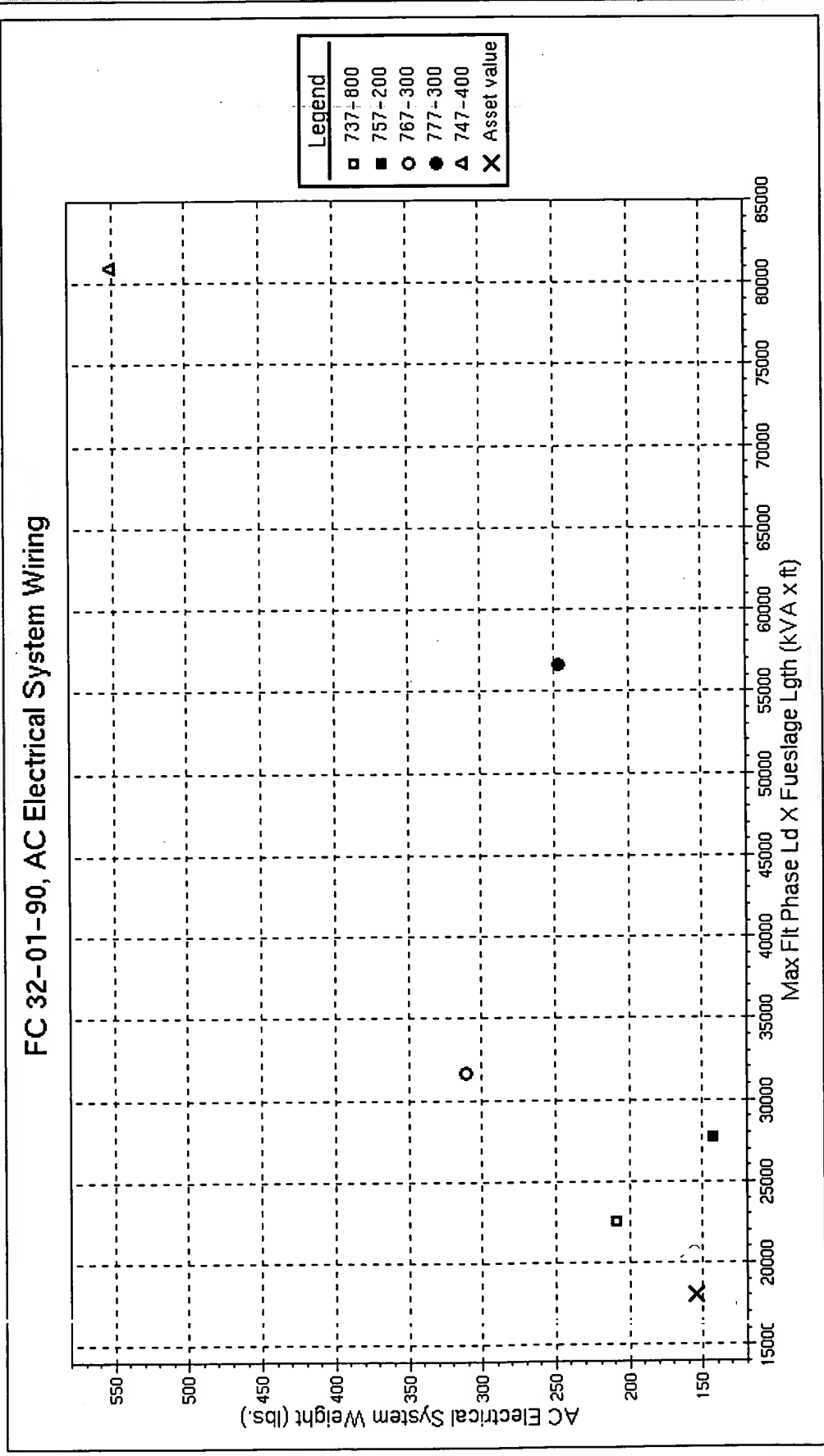
ASST: Chart

FC 32-08, Power Panels + ELMS



Return

106

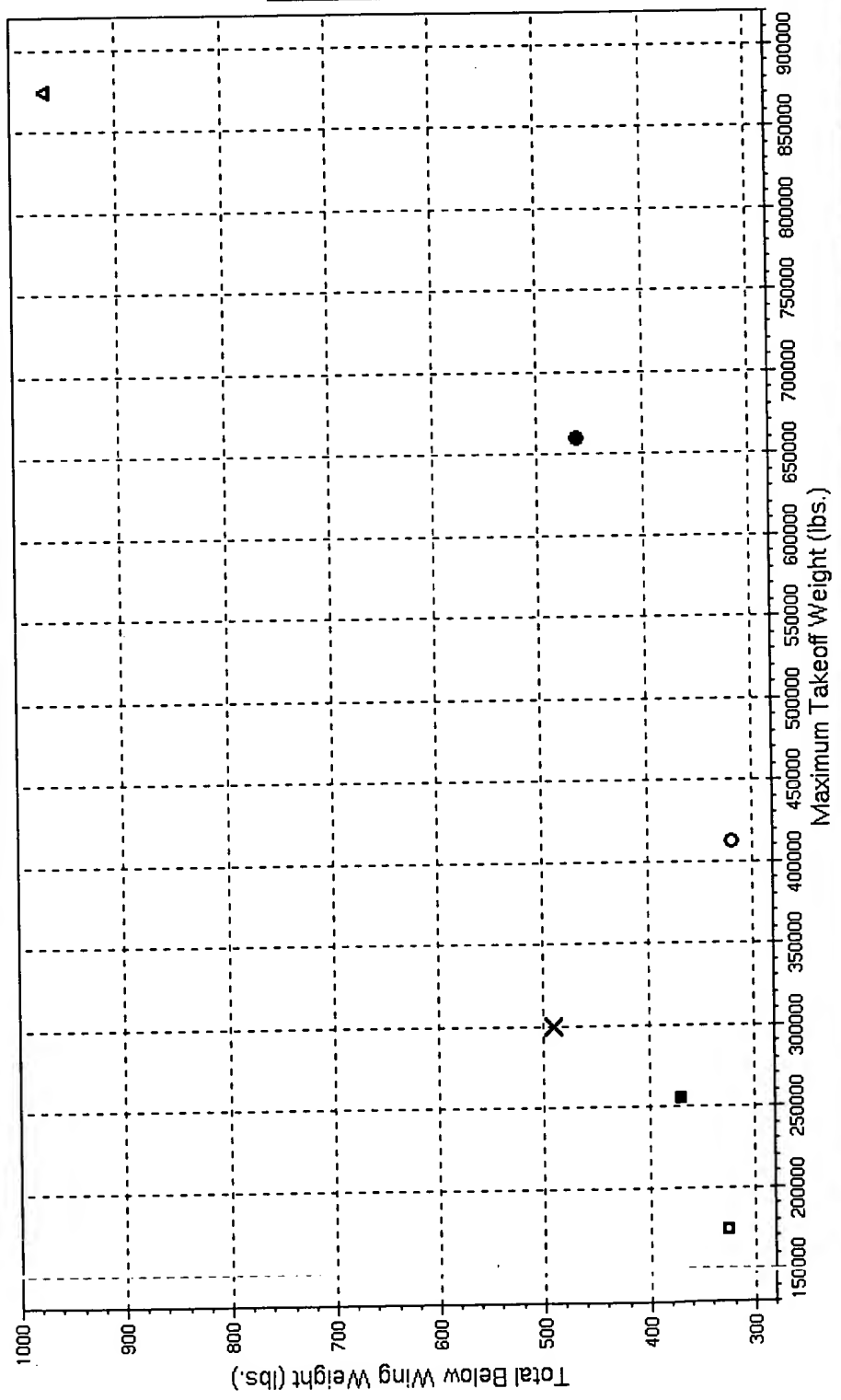


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108

FIGURE 80

Below Wing Weight



110

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Figure 81